

Structural health monitoring by combining machine learning and dimensionality reduction techniques

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Abstract

Structural Health Monitoring is of major interest in many areas of structural mechanics. This paper presents a new approach based on the combination of dimensionality reduction and data-mining techniques able to differentiate damaged and undamaged regions in a given structure. Indeed, existence, severity (size) and location of damage can be efficiently estimated from collected data at some locations from which the fields of interest are completed before the analysis based on machine learning and dimensionality reduction techniques proceed.

OPEN ACCESS

Published: 08/02/2019

Accepted: 10/12/2018

Submitted: 10/02/2018

DOI:
10.23967/j.rimni.2018.12.001

Keywords:
Non Destructive Testing
Machine Learning
Dimensionality Reduction

1. Introduction

Structural deterioration and degradation are of great concern worldwide, being damage the main cause of structural failure. A special attention must be paid in order to avoid the sudden failure of structural components. The improvements in the fields of low-cost displacement and acceleration transducers, signal conditioning and sampling hardware, electronic data acquisition systems, pushed the interest of the scientific community in the use of the dynamic response of structural systems as a tool to evaluate damage and safety. Recently, various non-destructive techniques based on changes in the structural vibrations patterns have been extensively published not only to detect the presence of damage but also to identify the location and the severity of it. Non-destructive Testing (NDT) methods are one of the most important topics in the Structural Health Monitoring (SHM) field. These methods have to be able to identify damage when it appears and capture and locate the damaged area.

Several system identification techniques exist to obtain unknown structural parameters as damping ratio, natural frequencies or mode shapes [1,2]. The basis of these methods is to extract information from some measurements on the structure, as, for example, accelerations or displacements. A first classification divides the methods in frequency domain and time domain methods.

Frequency domain techniques have the advantage of modal analysis methods, where the analysis can be done in some

range of frequencies of interest or with some structural modes. In [3,4,5,6,7] frequency domain responses are obtained from time series responses by non-parametric estimation and signal processing techniques which make use of the Fourier transform. Concerning modal analysis methods [8,9,10,11,12,13,14] they are based on the use of modal informations extracted from input-output measurements by means of the modal analysis methods or from only output data measured under the ambient excitation (wind, traffic loads, etc.) without making use of artificial forces.

Time domain methods avoid problems as leakage or closeness to natural frequencies. In [2, 15,16,17,18,19,20] authors identified modal parameters from time domain measurements and used the extracted vibration features and modal properties for detecting damage occurrence and/or location by comparing the identified modal properties with the original values. It is also possible with these methods [21,22,23] to directly detect damage based on the measured data. Another approach in this group [24,25,26] makes use of many signal processing techniques and artificial intelligence as analysis tools to investigate the vibration signals and extract features to represent the signal characteristics.

Another classification could be done depending on the nature of the excitation force: some methods work with a known impulse force [27], others work with unknown natural excitations [28], and a third group works with a combination of the previous ones [29].

Wavelet analysis is also an appealing technique widely used for the Non-Destructive Testing, in which a wavelet transform is applied on modal shapes of vibration. Since this analysis is capable to identify changes in the modal shapes, damage can be easily identified as well as its spatial location. There is a vast literature on the extensive use of wavelets [30,31,32,33,30,34,35,36,37].

Reduced Order Modeling (ROM) techniques have been widely used in order to locate damage under real-time constraints. In the field of SHM there exists lots of approaches making use of reduced order modeling. In [38] authors applied the Proper Orthogonal Decomposition (POD) to track the structural behavior followed by an improved particle filtering strategy (extended Kalman updating). Machine learning is also helping for extracting the manifold in which the solutions of complex and coupled engineering problems are living. Thus, uncorrelated parameters can be efficiently extracted from the collected data coming from numerical simulations, experiments or even from the data collected from adequate measurement devices. The Proper Orthogonal Decomposition (POD), that is equivalent to Principal Components Analysis (PCA), can be viewed as an information extractor from a data set that attempts to find a linear subspace of lower dimensionality than the original space. Moreover, PCA-based transformations preserve distances, where other nonlinear dimensionality reduction strategies fail to accomplish it. In [39] authors proposed a data-driven methodology for the detection and classification of damages by using multivariate data driven approaches and PCA. Support Vector Machine (SVM) was used for damage detection in [40].

It is also usual in this context approaches that combine Machine Learning Techniques and Reduced Order Modeling, like in [41] where authors used machine-learning algorithms to generate a classifier that monitors the damage state of the system and a Reduced Basis method to reduce the computational burden associated with model evaluations. Proper Orthogonal Decomposition approximations and Self-Organizing Maps (SOM) are combined to realize a fast mapping from measured quantities in order to propose a data-driven strategy to assist online rapid decision-making for an unmanned aerial vehicle that uses sensed data to estimate its structural state [42].

Data Mining techniques are also used in the context of the SHM. In [43] authors propose an approach for damage identification and optimal sensor placement in Structural Health Monitoring by using a Genetic Algorithm technique (GA) whereas in [44] authors combined Data Mining (GA), Machine Learning (PCA) and Deep Learning (Neural Networks) techniques in the damage identification context. Concerning Deep Learning techniques, it is interesting the work developed in [45] in which a smart monitoring of aeronautical composites plates based on electromechanical impedance measurements and artificial neural networks is presented. At its turn [46] proposes the same technique in the monitoring of a frame structure model for damage identification.

This paper proposes a new strategy based on the combination of model order reduction, that extracts a reduced basis from undamaged snapshots, that will serve for projecting any measured solution on it, with data-mining techniques. When projecting into this reduced basis the measured field, undamaged regions are expected being better approximated than the ones in which damage occurs. Thus, data-mining strategies could be then used to differentiate both regions (undamaged and damaged). Finally, in order to limit the number of points at which data is collected, the just described methodology is combined with a data-completion strategy based on the use of dictionary learning.

After this short introduction, next section addresses the data generator based on the solution of a elastodynamic model in a plate. Then, next section applies different techniques on the generated data in order to clusterize damaged and undamaged zones. Finally the same procedure is repeated but on the completed data obtained from data sparsely collected.

2. Elastodynamic model

The problem taken into consideration is depicted in Figure 1. The geometrical and mechanical properties of the square domain $\Omega = [0, L] \times [0, L]$ are defined in Table 1.

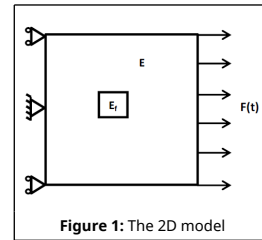


Table 1. Model parameters.

L : Length (m)	1
E : Young modulus (N/m^2)	210^{11}
ν : Poisson coefficient	0.25

A linear elastic behavior is assumed in the undamaged area, so that the relation between the stress σ and the strain ϵ reads

$$\sigma = \mathbb{C} \epsilon, \quad (1)$$

where \mathbb{C} is the Hooke's fourth order tensor. The relation between strain ϵ and displacement u writes

$$\epsilon = \nabla_s u, \quad (2)$$

where $\nabla_s \cdot$ is the symmetric gradient operator.

On the right boundary of the domain a traction is enforced, $F(t) = A \sin(\omega t)$, where $A = 10^6$ and $\omega = 2\pi 10^3$. Considering an isotropic material, plane stress conditions and using the Voigt notation, the Hooke's tensor can be written as

$$\mathbb{C} = \frac{E}{1 - \nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & (1 - \nu)/2 \end{bmatrix} \quad (3)$$

and the relation (1) as

$$\begin{bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{xy} \end{bmatrix} = \frac{E}{1 - \nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & (1 - \nu)/2 \end{bmatrix} \begin{bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \gamma_{xy} \end{bmatrix}. \quad (4)$$

The material is assumed homogeneous and isotropic everywhere, with degraded mechanical properties in the damaged region, with the Young modulus reduced by one order of magnitude, i.e. $E_f = E/10$. Moreover a non-linear behavior is prescribed in the damaged area. The particular choice of this nonlinear dependency is irrelevant, the important point being the fact that nonlinearities generate frequencies different to the one(s) involved in the loading of major relevance for identifying damage. For this reason in the sequel we consider the simplest nonlinear behavior in the damaged zones

$$\sigma = C(\varepsilon) \varepsilon. \quad (5)$$

The displacement field evolution $\mathbf{u}(\mathbf{x}, t)$ for $\mathbf{x} \in \Omega$ and $t \in I = [0, T]$ is described by the linear momentum balance equation

$$\rho \ddot{\mathbf{u}}(\mathbf{x}, t) = \text{div } \sigma, \quad (6)$$

where ρ is the density (kg/m^3).

The boundary $\partial\Omega$ is partitioned into Dirichlet, Γ_D , and Neumann, Γ_N , boundaries, where displacement and tractions are enforced respectively, as sketched in Figure 1. Without loss of generality homogeneous initial conditions $\mathbf{u}(\mathbf{x}, t = 0) = 0$ and $\dot{\mathbf{u}}(\mathbf{x}, t = 0) = 0$ are assumed.

The problem weak form associated with the strong form (6) lies in looking for the displacement field \mathbf{u} verifying the initial and Dirichlet boundary conditions such that the weak form

$$\rho \int_{\Omega} \ddot{\mathbf{u}} \cdot \mathbf{v} d\mathbf{x} + \int_{\Omega} \varepsilon(\mathbf{v}) \cdot (C(\varepsilon) \cdot \varepsilon(\mathbf{u})) d\mathbf{x} = \int_{\Gamma_N} \mathbf{F}(t) \cdot \mathbf{v} d\mathbf{x} \quad (7)$$

applies for any test function \mathbf{v} , with the trial and test fields defined in appropriate functional spaces.

For discretizing the weak form we introduce a standard explicit time-marching method in the time interval defined by T ($T = 0.005$ s in the numerical examples addressed later) and a time step Δt ($\Delta t = 10^{-6}$ s), with $t_{k+1} = (k+1)\Delta t$

$$\rho \int_{\Omega} \frac{\mathbf{u}^{k+1} - 2\mathbf{u}^k + \mathbf{u}^{k-1}}{\Delta t^2} \cdot \mathbf{v} d\mathbf{x} + \int_{\Omega} \varepsilon(\mathbf{v}) \cdot (C(\varepsilon^k) \cdot \varepsilon(\mathbf{u}^k)) d\mathbf{x} = \int_{\Gamma_N} \mathbf{F}^k \cdot \mathbf{v} d\mathbf{x}, \quad (8)$$

where the notation $\mathbf{u}(\mathbf{x}, t_k) = \mathbf{u}^k$ has been used.

The displacement field \mathbf{u} is then computed using a FEM space discretization with linear element over a uniform triangular mesh composed of 51×51 nodes such that the damaged area contains 128 elements. Using (2), (4) and (7), we obtain the discrete system [47]

$$\mathbf{M} \frac{\mathbf{u}^{k+1} - 2\mathbf{u}^k + \mathbf{u}^{k-1}}{\Delta t^2} + \mathbf{K}(\mathbf{u}^k) \mathbf{u}^k = \mathbf{f}^k, \quad (9)$$

where \mathbf{M} is the mass matrix, \mathbf{K} the stiffness matrix and $\mathbf{f}(t)$ the force vector.

3. Damage location via Machine Learning techniques

As discussed in the introduction, the main aim of this paper is proposing a strategy based on the combination of model order reduction based on PCA – Principal Component Analysis – that extracts a reduced basis from undamaged snapshots (that will serve for projecting any measured solution on it) with data-mining techniques.

Principal Components Analysis can be viewed as an information extractor from a data set that attempts to find a linear subspace of lower dimensionality than the original space. If the data has more complicated structures which cannot be well represented in a linear subspace, standard PCA fails for performing dimensionality reduction. In that case its nonlinear counterparts (kernel-based PCA or local-PCA) could be valuable alternatives

for defining reduced bases.

When projecting into this reduced basis the measured field, undamaged regions are expected being better approximated than the ones in which damage occurs. Thus, data-mining strategies could be then used to differentiate both regions (undamaged and damaged depicted in Figure 1).

For that purpose, problem (7) is solved with the same geometrical and mechanical properties defined in Table 1 but without any damaged zone.

By solving the discret equation (9) we obtain the undamaged displacement field $\mathbf{u}(\mathbf{x}, t)$ and from it the displacement field norm $w(\mathbf{x}, t)$ at the nodes \mathbf{x}_i of the spatial mesh at times $t_m = m \cdot \Delta t$, with $i \in [1, \dots, N]$ and $m \in [0, \dots, M]$. In the sequel we use the notation $w(\mathbf{x}_i, t_m) \equiv w_i^m$, and \mathbf{w}^m represents the vector of nodal values w_i^m at time t_m .

Then we apply the POD (equivalent to the PCA) to identify the most typical structure $\phi(\mathbf{x})$ among these \mathbf{w}^m , $\forall m$.

For that purpose we first define the matrix \mathbf{Q}_{ud} (where the subscript $_{ud}$ makes reference to its undamaged nature) from

$$\mathbf{Q}_{ud} = \begin{pmatrix} w_1^1 & w_1^2 & \dots & w_1^M \\ w_2^1 & w_2^2 & \dots & w_2^M \\ \vdots & \vdots & \ddots & \vdots \\ w_N^1 & w_N^2 & \dots & w_N^M \end{pmatrix} \quad (10)$$

and the two point correlation matrix \mathbf{C}

$$C_{ij} = \sum_{m=1}^M w_i^m(\mathbf{x}_i) w_j^m(\mathbf{x}_j), \quad (11)$$

or

$$\mathbf{C} = \sum_{m=1}^M \mathbf{w}^m \cdot (\mathbf{w}^m)^T = \mathbf{Q}_{ud} \cdot \mathbf{Q}_{ud}^T, \quad (12)$$

and then, within the usual POD framework, solve the resulting eigenvalue problem for obtaining the searched modes,

$$\mathbf{C}\phi = \alpha\phi, \quad (13)$$

where the i -entry of vector ϕ corresponds to $\phi(x_i)$.

In order to obtain a reduced-order model we select the P eigenvectors associated with the P largest eigenvalues, for example the ones greater than $\alpha_1 10^{-6}$. In many applications, the magnitude of the eigenvalues decreases very fast, fact that reveals that the solution \mathbf{w}^m can be approximated $\forall m$ from a reduced number P ($P \ll N$) of modes (eigenvectors).

In what follows we consider only the first two eigenvectors, because as explained later our goal is not to reconstruct the undamaged displacement but only differentiate between damaged and undamaged solutions, and our feeling is that the undamaged displacement is better represented in the reduced basis composed of the two modes extracted from the undamaged structure than the displacement associated with damaged zones.

For this purpose, we first select the firsts eigenmodes that are expected better representing solutions at the undamaged than

at the damaged regions. Note that the more eigenmodes are considered the less contrasted will be solutions in undamaged and damaged zones. In our numerical experiments we select the first two modes and we define matrix $\mathbf{B} = [\phi_1, \phi_2]$

$$\mathbf{B} = \begin{pmatrix} \phi_1(\mathbf{x}_1) & \phi_2(\mathbf{x}_1) \\ \phi_1(\mathbf{x}_2) & \phi_2(\mathbf{x}_2) \\ \vdots & \vdots \\ \phi_1(\mathbf{x}_N) & \phi_2(\mathbf{x}_N) \end{pmatrix}. \quad (14)$$

Now we repeat simulations, but now including the damaged zone obtaining, as before, the matrix of displacement norms \mathbf{Q}_d as

$$\mathbf{Q}_d = \begin{pmatrix} w_1^1 & w_1^2 & \dots & w_1^M \\ w_2^1 & w_2^2 & \dots & w_2^M \\ \vdots & \vdots & \ddots & \vdots \\ w_N^1 & w_N^2 & \dots & w_N^M \end{pmatrix}. \quad (15)$$

Now, the damaged solutions are projected onto the two-modes basis \mathbf{B} related to the undamaged structure, that results in

$$\beta = \mathbf{B}^T \mathbf{Q}_d, \quad (16)$$

from which the reconstructed damaged displacement norms result from

$$\mathbf{Q}_d^{rec} = \mathbf{B} \beta. \quad (17)$$

The residual between the real and reconstructed damaged displacement norms reads

$$\mathbf{R} = \mathbf{Q}_d - \mathbf{Q}_d^{rec}. \quad (18)$$

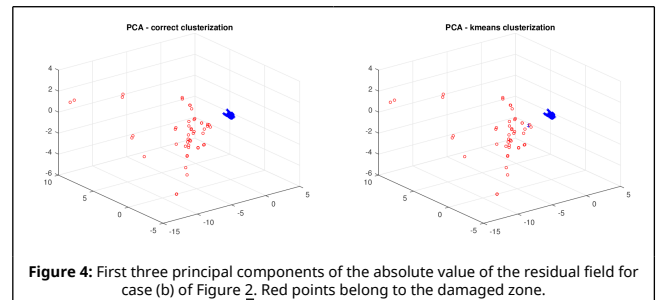
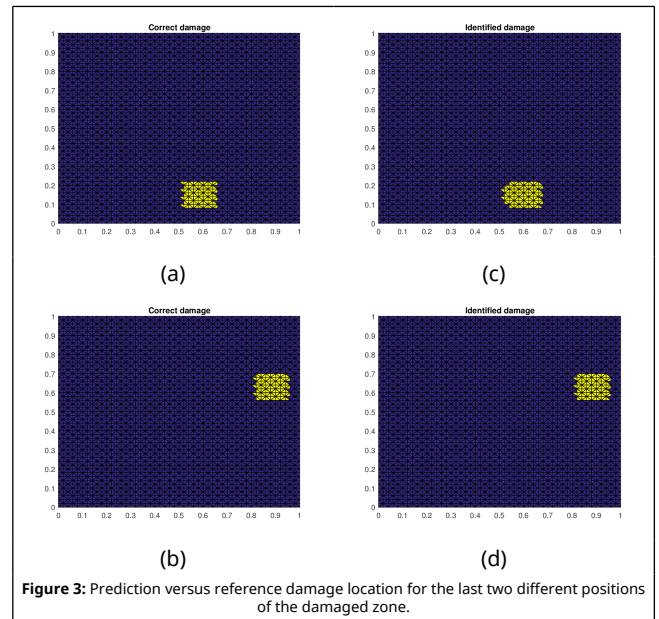
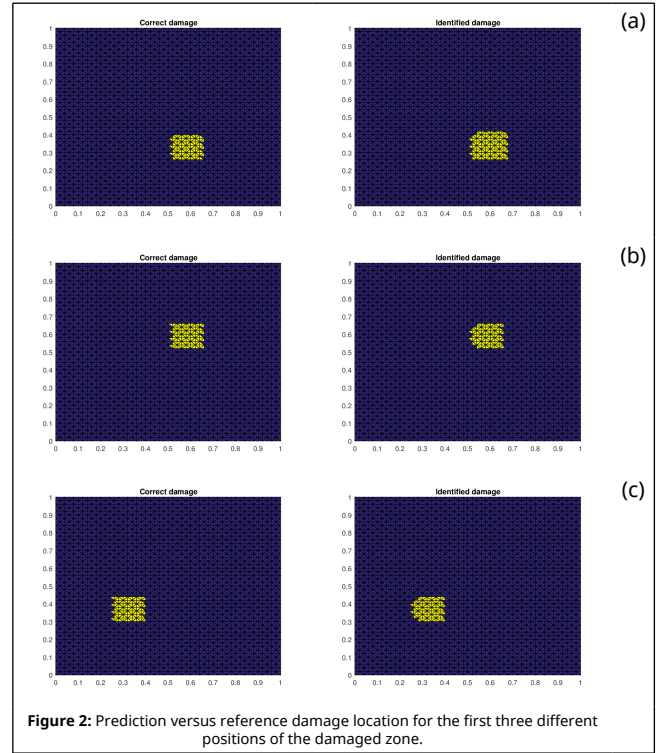
At this point a clustering technique is applied on the absolute value of the residual field \mathbf{R} . In this work the k-means strategy has been used. It proceeds in three steps:

1. An initial partition is done with two populations, i.e. $k = 2$. Many different methods could be used to choose initial centers of mass and a comparison of them is described in [48];
2. Each point is assigned to the cluster whose center of mass is closer;
3. Centers of mass are updated.

The second and third steps repeat until reaching a stable position of both centers of mass.

In order to reduce the dimensionality before applying the clustering, PCA is applied on the absolute value of the residual vectors.

The results of the damaged zone predicted by the proposed method for different positions of the damage are presented in Figures 2 and 3. We can see how it detects quite precisely the position of damaged regions. Moreover in Figure 4 one can see how even if the reduction of dimension performed by the PCA is extreme (only the three first principal components are taken) the zones (damaged and undamaged) are perfectly differentiated.



4. Data completion

The main difficulty when considering the approach discussed above is that the displacement field is needed in as many locations as possible (e.g. the nodes considered in the finite element mesh). Having access to all this local information could become prohibitive in practical applications. Thus, in this section, we consider data acquisition in few locations, from which fields are completed before applying the rationale discussed previously.

For this purpose a dictionary of simulations is performed, that contains the displacements and residuals fields everywhere for the damage located in different zones and taking different sizes (both perfectly known). In what follows we consider a subdivision of the domain like the one depicted in Figure 5.

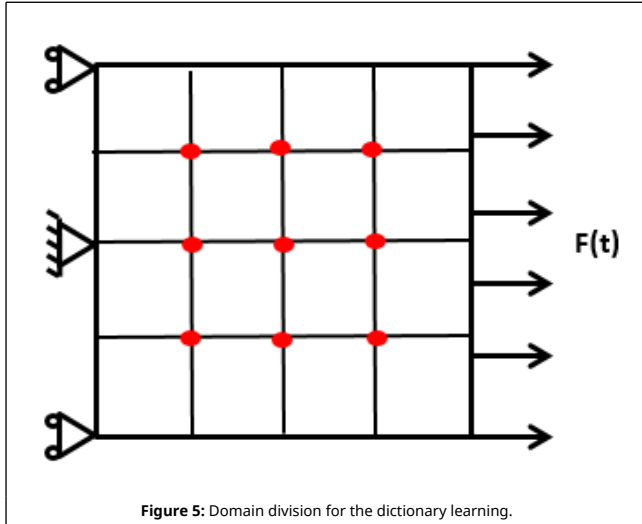


Figure 5: Domain division for the dictionary learning.

We consider 102 scenarios to create our dictionary: the first 48 are given by choosing as damaged zone one of the 16 areas depicted in Figure 5 and assigning to each area 3 different sizes of the damaged zone. The others 54 scenarios are given by choosing as damaged area the 9 intersections of the previous areas (the red points in Figure 5) and assigning to each point 6 possible different sizes of the damaged zone centered on it.

Obviously the dictionary can be enriched with many other locations and sizes of the damaged zone, but the goal in this work is to show how the dictionary learning technique can be used to perform data completion and for this reason the scenarios previously described seem sufficient.

Once the dictionary has been created we suppose that a displacement field related to an unknown damaged scenario is known at few locations, that is, at the positions where sensors are placed. In this work we suppose displacements accessible at the nodes of a 9×9 uniform grid. It is important to note that coarser dictionaries require much less sensors, whereas rich dictionaries require many measurements in order to identify the closest scenario. In practice we could proceed with coarser representations for online monitoring and richer representations for maintenance operations.

The displacement norms at those locations allow defining matrix $\tilde{\mathbf{Q}}_d$, where $\tilde{\mathbf{Q}}_d$ comes from \mathbf{Q}_d defined in Eq.(15) by taking the rows corresponding to the sensor points. Then we compute the residual at the sensor points using the same rationale that was considered in Eqs.(16) and (18), but with the difference that now coefficients β are obtained in a least-squares sense from

$$\beta = \arg \min_{\beta} ||\tilde{\mathbf{B}}\beta - \tilde{\mathbf{Q}}_d||_2, \quad (19)$$

where $\tilde{\mathbf{B}}$ has been obtained from \mathbf{B} defined in Eq.(14) by taking the rows corresponding to the sensor points. Let's note that Eq.(19) can be solved because the number of sensor points is greater than the number of functions in the basis \mathbf{B} (two in the present analysis). Then we reconstruct the damaged displacement norm field at the sensor points by computing

$$\tilde{\mathbf{Q}}_d^{rec} = \tilde{\mathbf{B}}\beta. \quad (20)$$

The residual between the reference and the reconstructed damaged displacement norm at the sensor points reads

$$\tilde{\mathbf{R}} = \tilde{\mathbf{Q}}_d - \tilde{\mathbf{Q}}_d^{rec}. \quad (21)$$

Then, we compute the gaps between $\tilde{\mathbf{R}}$ and the residual field at the sensor points of all the simulations in the dictionary, and we select as reference simulation the one that minimizes the norm of this error.

The complete residual field related to the reference simulation is noted by \mathbf{R}_{ref} . The reduced base \mathbf{G} used for data completion is then computed performing a POD on \mathbf{R}_{ref} that results in

$$\mathbf{G} = \begin{pmatrix} \psi_1(\mathbf{x}_1) & \psi_2(\mathbf{x}_1) & \dots & \psi_F(\mathbf{x}_1) \\ \psi_1(\mathbf{x}_2) & \psi_2(\mathbf{x}_2) & \dots & \psi_F(\mathbf{x}_2) \\ \vdots & \vdots & \ddots & \vdots \\ \psi_1(\mathbf{x}_N) & \psi_2(\mathbf{x}_N) & \dots & \psi_F(\mathbf{x}_N) \end{pmatrix}, \quad (22)$$

where F is the number of functions selected to compose the reduced base used for the completion.

In order to compute the complete residual field \mathbf{R}_{com} we compute first

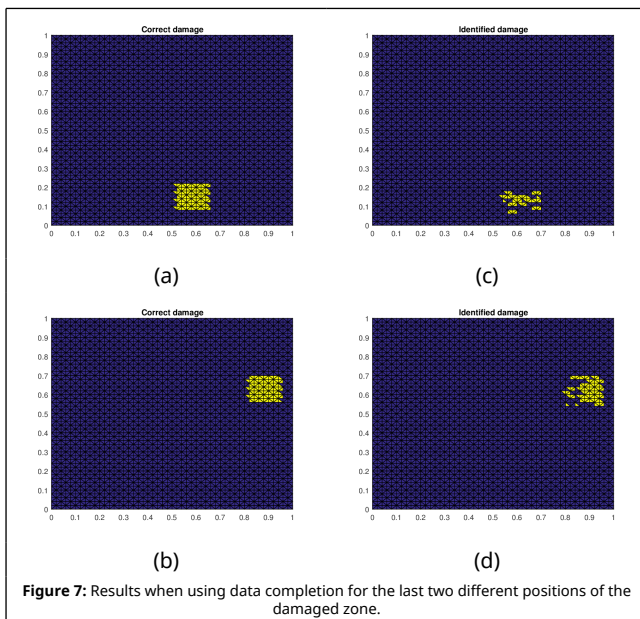
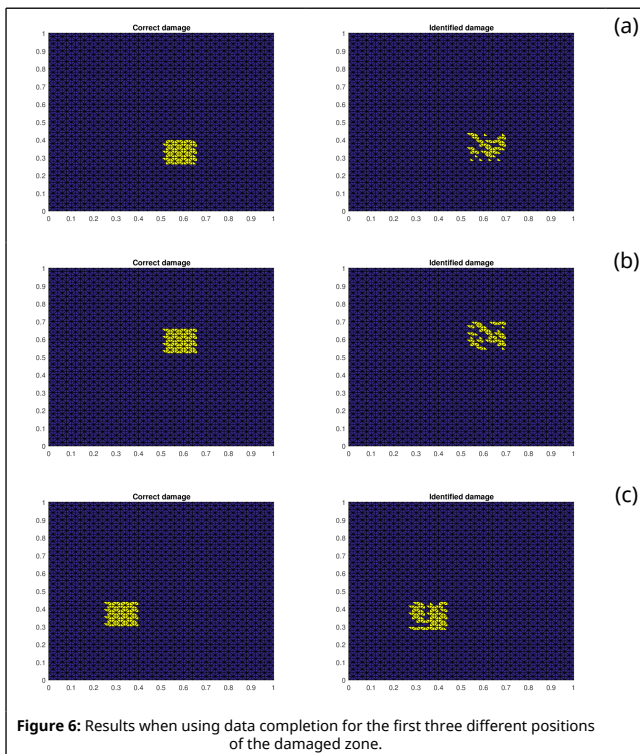
$$\gamma = \arg \min_{\gamma} ||\tilde{\mathbf{G}}\gamma - \tilde{\mathbf{R}}||_2, \quad (23)$$

where $\tilde{\mathbf{G}}$ has been obtained from \mathbf{G} defined in Eq.(22) by taking the rows corresponding to the sensor points and where F is smaller than the number of sensors. Thus, it results

$$\mathbf{R}_{com} = \mathbf{G}\gamma. \quad (24)$$

As in the previous section the k-means technique is then applied on the absolute value of the completed residual field \mathbf{R}_{com} . Again, in order to reduce the dimensionality of the problem, the PCA is applied on the absolute value of the completed residual field before performing clusterization.

Results on the damaged zone detection by using the proposed methodology for the different positions of the damage considered in the previous section are presented in Figures 6 and 7. We can notice that results are in good agreement to the ones presented in the previous section, with a quite good identification of the damaged zone.



5. Conclusions

In this paper we proposed a new efficient technique for real-time evaluation of damage in structures. For that purpose few POD modes associated with the undamaged structure were used for reconstructing the fields of interest. As expected, as soon as damage occurs, the projection into the undamaged modes allows differentiating using standard clustering techniques, damaged and undamaged regions. Moreover, to avoid data collection on the whole structure, a procedure for collecting data at few specified locations was proposed. Then from the collected data at these points, the fields of interest where completed everywhere, allowing for an accurate damage location.

The numerical test performed proved the validity and potential of the proposed approach that should be now validated experimentally.

Acknowledgements

The research leading to this work was supported by the ESI group chairs at Centrale Nantes and ENSAM ParisTech and it has received funding from the European Union's Horizon 2020 research and innovation programme under the Marie Skłodowska-Curie grant agreement [675919].

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